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# NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

TECHNICAL NOTE

No. 1507

SYMBOLS FOR COMBUSTION RESEARCH

By NACA Subcommittee on Combustion

Flight Propulsion Research Laboratory  
Cleveland, Ohio

**FOR REFERENCE**

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NOT TO BE TAKEN FROM THIS ROOM



Washington  
June 1948

LANGLEY MEMORIAL AERONAUTICAL  
LABORATORY  
Langley Field



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## INTRODUCTION

Because the NACA Subcommittee on Combustion recognizes the need for standard combustion symbols to facilitate the exchange of technical information and to provide a basis for comparison of results obtained by various investigators, such a list of symbols has been prepared. An attempt has been made to be consistent with long accepted usage in the fields of thermodynamics and aerodynamics.

These symbols for combustion research were prepared by the Panel consisting of Dr. Ernest F. Flock, chairman, Dr. Bernard Lewis, Mr. L. Richard Turner, Dr. Stewart Way, and Professor Glenn C. Williams. They were recommended to the Power Plants Committee of the NACA by its Subcommittee on Combustion at a meeting of the subcommittee held September 25, 1946, and were approved by the Committee on Power Plants for Aircraft on May 12, 1947 and by the Executive Committee of the NACA on June 5, 1947. The membership of the NACA Subcommittee on Combustion at the time of the adoption of these symbols was as follows:

Professor Glenn C. Williams, Chairman, Massachusetts Institute of Technology  
Mr. G. L. Wander, Air Materiel Command, Wright Field  
Lt. Comdr. C. C. Hoffman, U. S. N. Bureau of Aeronautics  
Dr. Ernest F. Flock, National Bureau of Standards  
Dr. Bernard Lewis, Bureau of Mines  
Dr. W. T. Olson, NACA, Cleveland Laboratory  
Mr. A. M. Rothrock, NACA  
Dr. W. G. Berl, Johns Hopkins University  
Mr. A. J. Nerad, General Electric Company  
Professor Robert N. Pease, Princeton University  
Dr. William J. Sweeney, Standard Oil Development Company  
Dr. Stewart Way, Westinghouse Electric Corporation

## SYMBOLS, DIMENSIONS, AND TYPICAL UNITS

<u>Symbol</u>	<u>Concept</u>	<u>Dimensions</u>	<u>Typical units</u>
A	area, cross sectional	$L^2$	sq ft
C	coefficient	none	none
$C_x$	concentration; moles of substance "x" per unit volume	$m/L^3$	lb moles/cu ft
c	velocity of sound $\left(\sqrt{\frac{\delta p}{\delta \rho}}\right)$	$L/t$	ft/sec
$c_p$	specific heat at constant pressure	$L^2/t^2T$	Btu/(lb)(°F)
$c_v$	specific heat at constant volume	$L^2/t^2T$	Btu/(lb)(°F)
D	diameter	L	ft
e	specific internal energy	$L^2/t^2$	Btu/lb
F	thrust; force	$mL/t^2$	lb
f	fuel-air ratio	none	none
g	standard acceleration of gravity	$L/t^2$	ft/sec <sup>2</sup>
h	specific enthalpy	$L^2/t^2$	Btu/lb
$h_o$	specific lower heat of combustion; ideal enthalpy change for isothermal constant-pressure combustion with product water remaining in vapor phase	$L^2/t^2$	Btu/lb
$h_t$	total specific enthalpy	$L^2/t^2$	Btu/lb
$h_v$	specific enthalpy of vaporization at constant pressure	$L^2/t^2$	Btu/lb

<u>Symbol</u>	<u>Concept</u>	<u>Dimensions</u>	<u>Typical units</u>
J	mechanical equivalent of heat	none	none
K	constant	none	none
L	length	L	ft
M	Mach number; also molecular weight	none	none
m	mass, total	m	slugs
n	polytropic exponent	none	none
p	absolute static pressure	$m/Lt^2$	lb/sq ft
$p_t$	absolute total pressure	$m/Lt^2$	lb/sq ft
Q	total quantity of heat transferred by conduction, convection, or radiation; also volume rate of flow	$mL^2/t^2$  $L^3/t$	Btu  cu ft/sec
q	dynamic pressure; one-half momentum flux per unit area	$m/Lt^2$	lb/sq ft
R	universal gas constant	$L^2/t^2T$	ft-lb/(°F)(mole)
r	pressure ratio	none	none
S	area, surface	$L^2$	sq ft
s	specific entropy	$L^2/t^2T$	Btu/(lb)(°F)
T	absolute static temperature	T	°R
$T_t$	absolute total temperature	T	°R
t	time	t	sec
V or u	velocity	$L/t$	ft/sec

<u>Symbol</u>	<u>Concept</u>	<u>Dimensions</u>	<u>Typical units</u>
$v$	specific volume	$L^3/m$	cu ft/lb
$W$	weight flow per unit time	$mL/t^3$	lb/sec
$\gamma$ or $k$	specific-heat ratio ( $c_p/c_v$ )	none	none
$\Delta$	finite change or difference	none	none
$\delta$	static pressure divided by NACA standard sea-level pressure	none	none
$\delta_t$	total pressure divided by NACA standard sea-level pressure	none	none
$\eta$	efficiency	none	none
$\theta$	static temperature divided by NACA standard sea- level temperature	none	none
$\theta_t$	total temperature divided by NACA standard sea- level temperature	none	none
$\lambda$	wave length	$L$	ft
$\mu$	absolute viscosity	$m/Lt$	lb sec/sq ft
$\nu$	frequency; also kinematic viscosity ( $\mu/\rho$ )	$L^2/t$	sq ft/sec
$\rho$	specific density	$m/L^3$	slugs/cu ft
$\sigma$	specific density divided by NACA standard sea-level density	none	none
$\tau$	ratio of two static tempera- tures; also, characteristic $t$ time of a thermocouple	none	none sec
$\tau_t$	ratio of two total tem- peratures	none	none

## SUBSCRIPTS

Numerical subscripts shall be used to indicate positions in an apparatus, process, or cycle. When used to indicate stations in a burner, the numbers assigned to the stations shall increase from inlet to exit.

a	air	f	fuel
ad	adiabatic	h	heat exchanger or intercooler
am	ambient	i	indicated
av	average	is	isentropic
b	burner or burned gas	j	jet or exhaust nozzle
c	compressor	n	nozzle or net
calc	calculated	sl	NACA standard sea level
corr	corrected	std	standard
cr	critical	t	total or turbine
d	diffuser or duct		

Flight Propulsion Research Laboratory,  
National Advisory Committee for Aeronautics,  
Cleveland, Ohio, March 11, 1948.